can be enhanced, e.g., by perforating the vane, or an alternate measurement scheme should be considered, e.g., the constrained vane discussed by Lenschow. 9

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Errata

Fatigue—A Test Integrated Damage Modeling Approach

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EQUATION (5) on page 567 of the subject paper should read as follows:

$$\frac{1}{N} \simeq \left(\frac{1}{N_{\text{approx}}}\right) = \lambda \left(\frac{1 - \kappa_o f_m}{f_a}\right)^{\beta_o}
+ \lambda_o \ln\left(\frac{1 - \kappa_o f_m}{f_a}\right) \left(\frac{1 - \kappa_o f_m}{f_a}\right)^{\beta_o} (\beta - \beta_o)
- \lambda_o \beta_o \frac{f_m}{f_a} \left(\frac{1 - \kappa_o f_m}{f_a}\right)^{\beta_o - 1} (\kappa - \kappa_o)$$
(5)

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Index categories: Aircraft Structural Design (Including Loads); Materials, Properties of; Aircraft Structural Materials.